

Using GPS to Determine Position Error Corrections

Purpose

- **To describe a pitot static Flight Test Technique**
 - ~ review earlier approaches to using GPS
 - ~ explain limitations of earlier methods
 - ~ show how limitations are alleviated
- **To describe the pros and cons of the proposed FTT**
- **To show test results that validate the proposed method**
 - ~ compare GPS results to traditional methods

Why use GPS for PEC?

- **The attraction**
 - ~ no aircraft modification required
 - » no trailing cone or aircraft plumbing mod
 - » no flight test boom
 - ~ no limitation on speed or altitude
 - » can be done down to near stall
 - » any altitude
 - ~ easy data reduction
 - » no correlation with pace aircraft, ground radar, or other references required

The Basic Idea

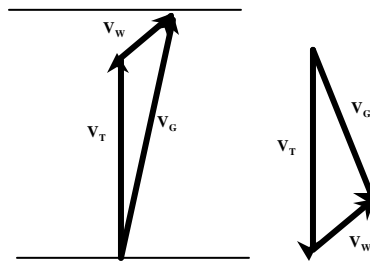
- **Using the precision of GPS**
 - ~ accurate ground speed cheaply and easily obtained in any aircraft
 - ~ get true airspeed from ground speed as in the traditional ground course FTT
 - » requires negating the effect of wind
 - ~ with true airspeed, can obtain speed position error correction (DV_{pc})

$$(V_i + DV_{ic} + DV_{pc} + DV_c)/S^{1/2} = V_{true}$$

- ~ note: still requires calibrated (preferably sensitive) airspeed indicator

Traditional Ground Course

- **To remove the effect of wind**
 - ~ fly a heading directly between two ground markers, allowing the aircraft to drift
 - ~ repeat 180° opposite heading
 - ~ use distance/time to get component of ground speed
 - ~ average the two ground speeds

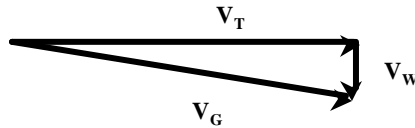


Old GPS Method 1

- **Do the same as traditional ground course method**
 - ~ but must fly directly into and away from the wind
 - ~ if not, no way to determine the components of ground speed
 - » unknown wind gives unknown contribution to V_g
- **Practical problem**
 - ~ determining wind direction
 - ~ done by flying until heading = track
 - ~ not sufficiently accurate for PEC determination

Old GPS Method 2

- Same as method 1, but
 - ~ fly directly perpendicular to the wind
 - ~ use average ground speeds and drift angle to solve for V_{true}



- Same problem as method 1
 - ~ if not directly perpendicular to the wind
 - » unknown effect on ground speed

Old GPS Method 3

- Same as traditional ground course method
 - ~ fly directly towards a GPS waypoint 8,000 nm away
 - ~ determine ground speed by timing GPS range changes
 - » this gives normal component of ground speed
 - no crab effect with remote waypoint
 - » average both towards and away components
- Practical problems
 - ~ Update of range on handheld GPS units is from 1 to 3 second interval
 - » okay only for range intervals that result in long times relative to timing errors

Old GPS Method 4

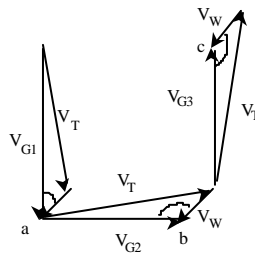
- **Same as traditional ground course method**
 - ~ fly true north or south
 - ~ determine ground speed by timing GPS *latitude* changes
 - » hack watch when whole latitude “rolls over”
 - minimizes update rate problems associated with range to a waypoint
 - » reverse course and repeat
- **Practical problems**
 - ~ minimal problems for low speed data points
 - ~ higher speeds requires longer runs, larger times
 - » possible wind change over large distances

Old GPS Method 5

- **Horseshoe track method (*Kitplanes*, Feb 95)**
 - ~ fly three legs with each perpendicular ground tracks, noting GPS ground speed on each
 - ~ determine true airspeed by solving three equations in three unknowns
- **Practical problem**
 - ~ only problem is the need to fly close to the ground, tracking perpendicular ground references

Old GPS Method 5

- **Horseshoe track method**
 - ~ fly three legs with each perpendicular tracks, noting GPS ground speed on each
 - ~ determine true airspeed by solving three equations in three unknowns



Old GPS Method 5

- **Solving three equations:**

$$\text{True airspeed : } V_T = \frac{1}{2} \sqrt{\left(V_1^2 + V_2^2 + V_3^2 + V_1^2 \times \left(\frac{V_3^2}{V_2^2} \right) \right)}$$

$$\text{Wind velocity } V_W = \sqrt{\left(\frac{V_1 - V_3}{2} \right)^2 + \left(\frac{V_2 - V_1 \times V_3 / V_2}{2} \right)^2}$$

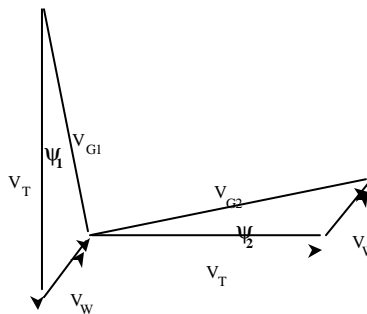
$$\text{Wind direction } \mathbf{y}_W = \tan^{-1} \frac{(V_2 - V_1 \times V_3 / V_2)}{(V_1 - V_3)}$$

Old GPS Method 6

- Almost all GPS units show both ground speed and ground track at the same time
- Using more information from accurate GPS
 - ~ less time spent taking data (more data/lb)
 - ~ less opportunity for pilot to be off condition
- Method
 - ~ fly two perpendicular headings
 - ~ record ground speed and track
 - ~ allows solving for wind direction, speed, and true airspeed

Old GPS Method 6

- If we know V_{G1} and V_{G2} and track angles ψ_1 and ψ_2
 - ~ Then we can write two equations in two unknowns, V_t and V_w
 - ~ With V_t and V_w we can also solve for wind direction



Old GPS Method 6

- Solving the two equations for the two leg speed and track method:

$$\text{True airspeed } V_T = \frac{V_{G_2}^2 - V_{G_1}^2}{V_{G_2} \cos(j_2) - V_{G_1} \cos(j_1)}$$

$$\text{Wind velocity } V_W = \left[V_{G_1}^2 + V_T^2 - 2V_TV_{G_1} \cos(j_1) \right]^{1/2}$$

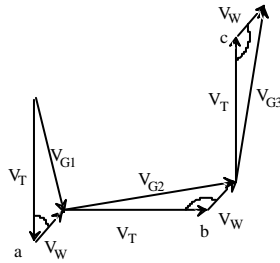
$$\text{Wind direction } y_W = \sin^{-1} \left[\frac{V_{G_1}}{V_W} \sin(j_1) \right]$$

Old GPS Method 6

- Practical problem
 - ~ accuracy and resolution of aircraft heading
 - ~ leads to large scatter in results
 - » 10 to 15 kts standard deviation typical
- Conclusion
 - ~ unsuitable for PEC

Proposed Method

- **Horseshoe heading method**
 - ~ fly three legs with each perpendicular headings, noting GPS ground speed on each
 - ~ determine true airspeed by solving three equations in three unknowns



Angle a = ψ
 Angle b = $90 + \psi$
 Angle c = $180 - \psi$

Equations

- **Solving the three equations for the horseshoe heading method:**

$$\text{Wind direction } \mathbf{y} = \tan^{-1} \left[\frac{-V_{G1}^2 + 2V_{G2}^2 - V_{G3}^2}{V_{G3}^2 - V_{G1}^2} \right]$$

$$\text{Wind velocity } V_W = \frac{1}{2} \left[V_{G3}^2 + V_{G1}^2 \pm \sqrt{(V_{G3}^2 + V_{G1}^2)^2 - \left(\frac{-V_{G1}^2 + 2V_{G2}^2 - V_{G3}^2}{\sin \mathbf{y}} \right)^2} \right]^{\frac{1}{2}}$$

$$\text{True airspeed } V_T = \sqrt{\frac{V_{G3}^2 + V_{G1}^2}{2} - V_W^2}$$

Software

- **“GPS True”**
 - ~ Windows 95 program
 - » available on <http://www.ntps.com>
 - ~ Inputs
 - » Initial heading
 - » V_{G1} , V_{G2} , and V_{G3}
 - ~ Assumptions
 - » left turns between stable points
 - » indicated speed, altitude, wind speed & direction, and temperature constant on all three points
 - ~ Output
 - » wind speed & direction, true airspeed
- Excel spreadsheet also available

“GPS True”

<u>F</u> ile <u>E</u> dit <u>H</u> elp	
Point Number <input type="text"/>	<input type="button" value="Calculate"/>
First Heading <input type="text"/>	Wind Speed <input type="text"/>
Ground Speed 1 <input type="text"/>	Wind Direction <input type="text"/>
Ground Speed 2 <input type="text"/>	True Airspeed <input type="text"/>
Ground Speed 3 <input type="text"/>	<input type="button" value="Next Point"/>

Excel Spreadsheet

		Heading	Vg1	Vg2	Vg3	Wind dir	Vwind	Vtrue GPS
	System	(deg mag)	(kts)	(kts)	(kts)	(deg)	(kts)	(kts)
1	Copilot	180	107.3	93.2	93.2	45	10.0	100.0
2	Duchess	270	126.0	143.0	138.0	332	12.9	131.5
3	Duchess	180	143.0	138.0	127.0	341	8.5	135.0
4	Duchess	90	138.0	127.0	126.0	310	7.8	131.9
5	Duchess	360	127.0	126.0	143.0	312	12.0	134.7
6	Duchess	270	135.0	152.0	143.0	344	14.2	138.3

- available on <http://www.ntps.com>

Validation Test

- The proposed FTT and data reduction were tried and the results compared to truth data
 - ~ aircraft: SW3 Merlin
 - » calibrated instruments
 - ~ GPS unit: Garmin 95
 - ~ truth data: trailing cone and Kiel tube
 - » truth data results: $s = 0.384$ kts from smooth curve fit
- The trial was conducted in conjunction with a two week introductory course on flight testing CAFB Cold Lake, Canada

Winds During the Trial

- The points were spread out over 3 sorties and nearly 2 hours
- Reduction of GPS data gave the following winds:

	Wind Speed	Wind Direction	Approx. Time of Day
1	17.4	283	9:30
2	15.5	286	9:40
3	13.5	285	10:15
4	12.5	286	11:00
5	1.0	120	11:10
6	11.3	252	11:00

- Point #5 was eliminated due to the unreasonable winds and unlikely deviation of PEC compared to other points

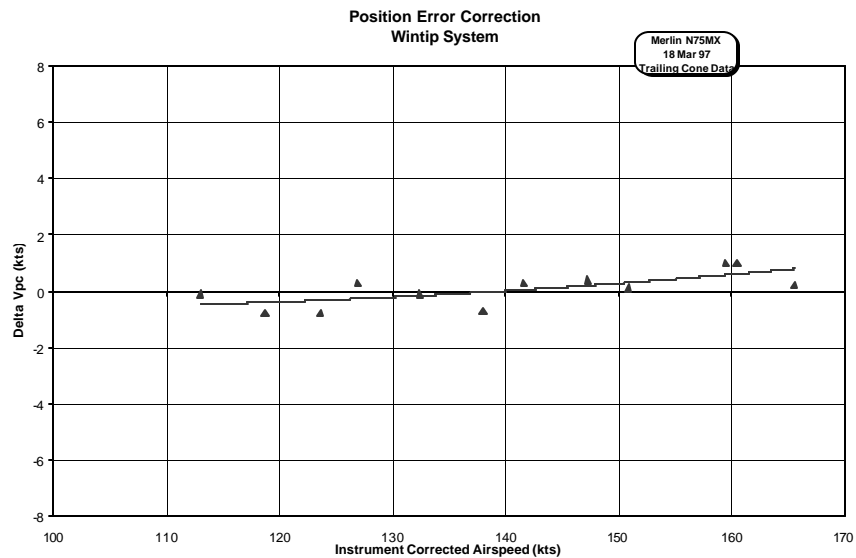
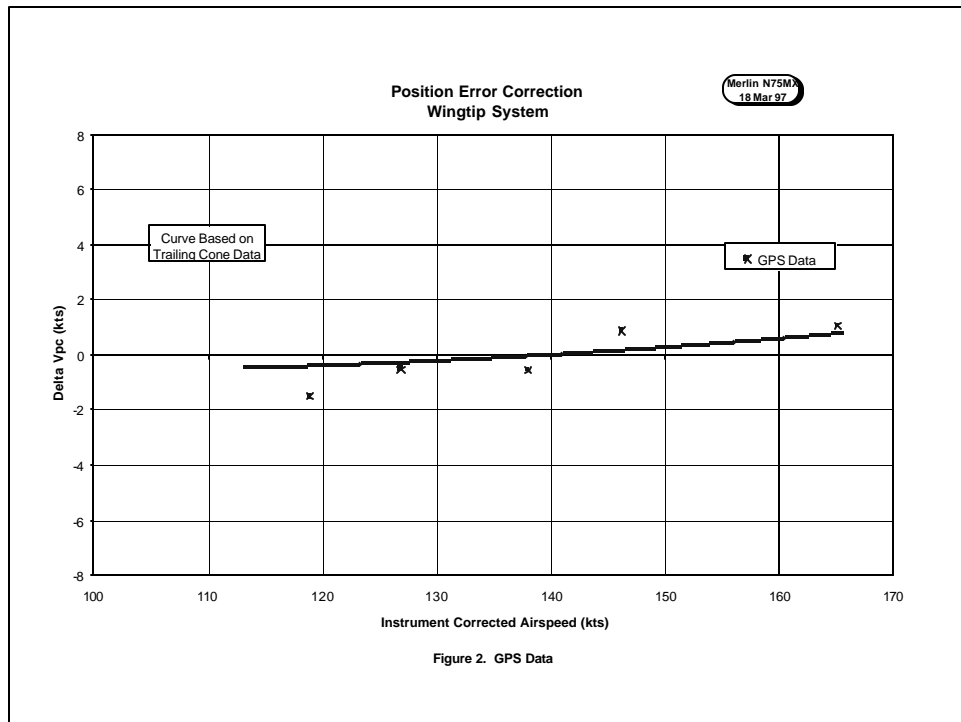


Figure 1. Trailing Cone Data



Differential GPS?

- **Trials were done with civilian code GPS**
- **Differential GPS or “P” codes would significantly improve accuracy**
 - ~ but accuracy of position
 - ~ not necessarily accuracy of ground speed
- **Most errors result in a bias that does not rapidly change with time**
 - ~ atmospheric effects
 - ~ DOP
 - ~ military dithering
- **GPS speed**
 - ~ if time differentiated position, unaffected by a constant bias
 - ~ if doppler, unaffected by position errors

Conclusions

- **“Horseshoe heading” method works well**
- **Generally excellent correlation**
 - ~ S of trailing cone data from best fit curve = 0.38 kts
 - ~ S of trailing GPS data from the same curve = 0.53 kts
 - ~ all ground course methods assume zero error in total pressure measurement
- **Proposed method should be applicable to a wide variety of aircraft**
 - ~ successfully used on
 - » several light aircraft
 - » Merlin, MB-326 (~0.6 M), C-130, SK-35 (~.9 M)